

Problem Set 12 (PS 12) Due Tuesday April 18

Problems 5.12, 5.16, 5.17 and 5.18 from the Anderson text

## 5.12

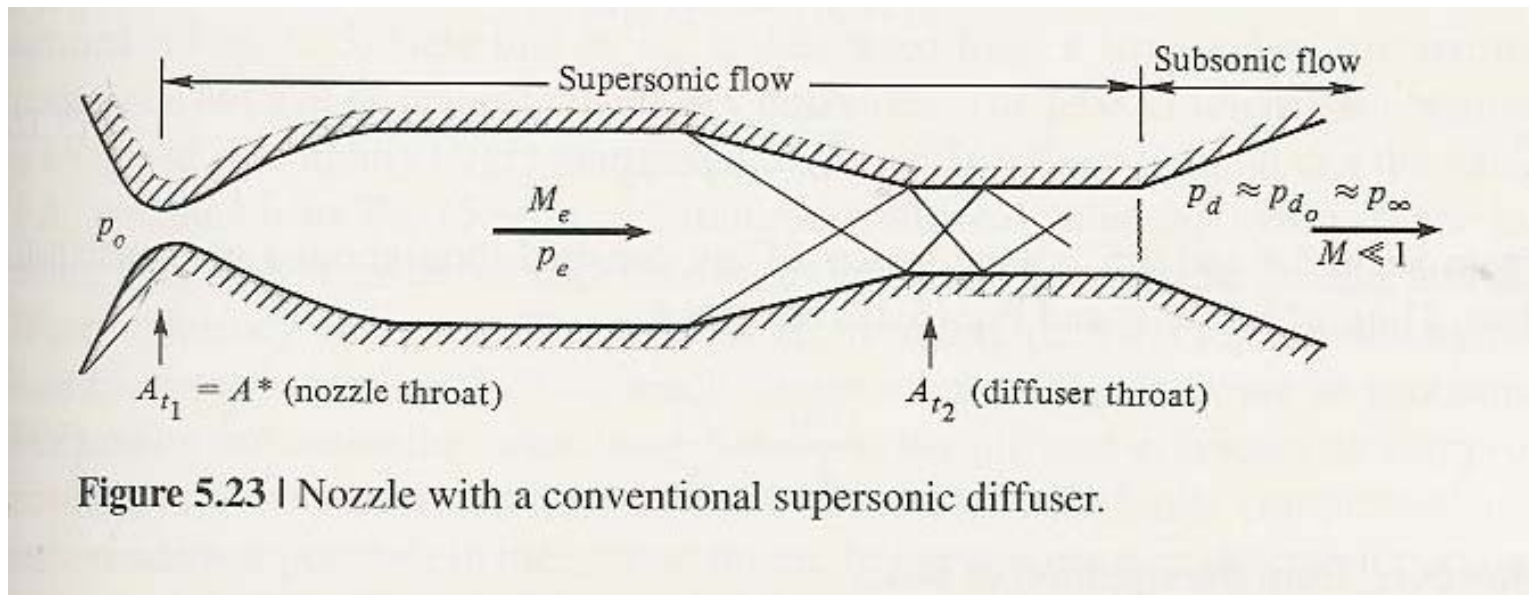


Figure 5.23 | Nozzle with a conventional supersonic diffuser.

Consider the tunnel to be completely isentropic,

$$\text{isentropic @ } \frac{A_e}{A_{t1}} = 104.2; \quad M_{\text{test}} = 7.$$

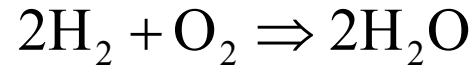
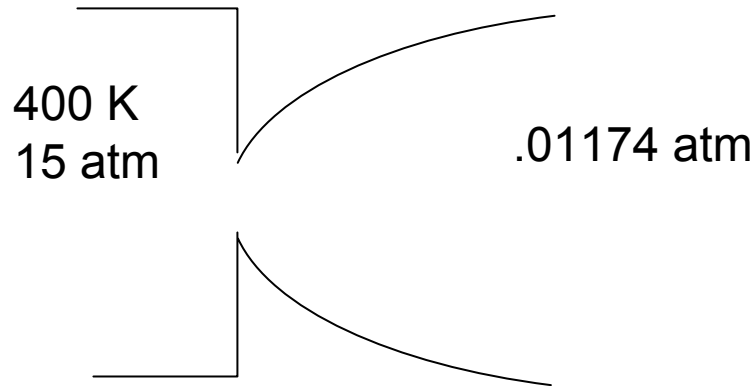
$$\left( \frac{A_{t2}}{A_{t1}} \right)_{\min} = \frac{p_{0t1}}{p_{0t2}} \quad (5.36)$$

for an normal shock at the end of the test section – the worst case

$$\text{norma shock @ } M_1 = 7; \quad \frac{p_{02}}{p_{01}} = .01535$$

$$\left( \frac{A_{t2}}{A_{t1}} \right)_{\min} = \frac{1}{.01535} = 65.15$$

5-16



assume an ideal gas

assume isentropic

assume 15 atm is stagnation

$$\frac{p_0}{p} = \left( 1 + \frac{\gamma - 1}{2} M^2 \right)$$

$$\left( \left( \frac{p_0}{p} \right)^{\frac{\gamma - 1}{\gamma}} - 1 \right) \frac{2}{\gamma - 1} = M^2$$

$$\left( \left( \frac{15}{.01174} \right)^{.1803} - 1 \right) \frac{2}{.22} = M^2$$

$$M = 4.9$$

## 5.17

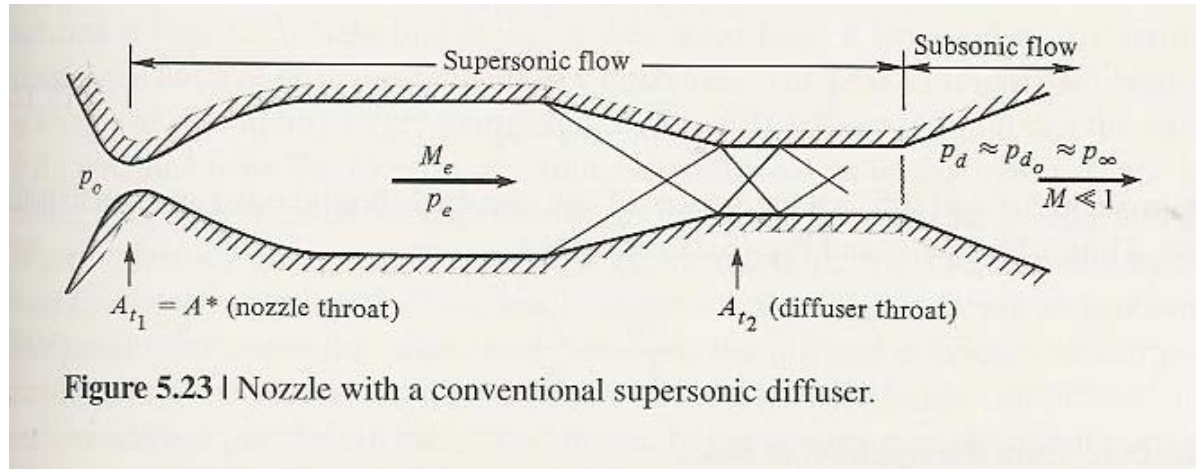


Figure 5.23 | Nozzle with a conventional supersonic diffuser.

a) isentropic @  $M = 3$ ;  $\frac{A}{A^*} = 4.235 = \frac{A_e}{A_{t1}}$

b) normal shock at diffuser

normal shock @  $M_1 = 3$ ;  $\frac{p_{O2}}{p_{O1}} = .3283$

equation (5.36)  $\frac{A_{t2}}{A_{t1}} = \frac{p_{O1}}{p_{O2}} = \frac{1}{.3283} = 3.046$

isentropic @  $M = 3$ ,  $\frac{p_0}{p} = 36.73$ ,  $\frac{T_0}{T} = 2.740$

c)  $p_0 = 36.73 \times .1 = 3.673 \text{ atm}$

d)  $T_0 = 2.74 \times 400 = 1096 \text{ K}$

## 5.18

Wind tunnels 3000 K test sections of  $M=10$  and  $M=20$

$$\text{isentropic @ } M = 10; \quad \frac{T_o}{T} = 21$$

$$T_{\text{test}} = \frac{3000}{21} = 142.86$$

$$a_{\text{test}} = \sqrt{\gamma RT}$$

$$a_{\text{test}} = \sqrt{1.4 \times 287 \times 142.86}$$
$$= 239.6 \text{ m/sec}$$

$$V_{\text{test}} = M \times a$$

$$V_{\text{test}} = 10 \times 239.6 = 2396 \text{ m/sec}$$

$$\text{isentropic @ } M = 20; \quad \frac{T_o}{T} = 81$$

$$T_{\text{test}} = \frac{3000}{81} = 37.04$$

$$a_{\text{test}} = \sqrt{\gamma RT}$$

$$a_{\text{test}} = \sqrt{1.4 \times 287 \times 37.04}$$
$$= 122. \text{ m/sec}$$

$$V_{\text{test}} = M \times a$$

$$V_{\text{test}} = 10 \times 122. = 2440 \text{ m/sec}$$

Air would liquefy before reaching 37 K

High  $m$  is from low temperature not high velocity